

# An Efficient and Robust $I$ - $f$ Control of Sensorless IPMSM With Large Startup Torque Based on Current Vector Angle Controller

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**Abstract**—Interior permanent magnet synchronous motor (IPMSM) with back electromotive-force (EMF) based sensorless field-oriented control (FOC) is widely used in medium-high speed applications. Unfortunately, the back EMF is too small to be estimated accurately during low speed operation. Hence, the current-frequency ( $I$ - $f$ ) control with controllable current vector is used for startup. However, the conventional  $I$ - $f$  control is hard to realize the tradeoff between startup speed and load capacity, suffering high-acceleration and high-load startup failure. In this article, a current vector angle controller for sensorless IPMSM startup is proposed to adjust speed-up-stage current acceleration and constant-speed-stage current amplitude with load automatically. The proposed angle controller guarantees the current vector angle close to the maximum torque per ampere angle, which contributes to large startup torque, high operating efficiency and smooth transition from  $I$ - $f$  startup to sensorless FOC. Furthermore, the robustness against parameters variation of angle controller is evaluated quantitatively. These theoretical advantages are verified in experiments with a 380 V, 1.5 kW IPMSM. In particular, the experimental results demonstrate that the proposed  $I$ - $f$  control realizes full-load startup even with  $\pm 50\%$  flux linkage and  $\pm 30\%$  inductance variation. Furthermore, the startup acceleration auto-adjusts with load to reconcile startup speed and load capacity.

**Index Terms**—Current vector angle controller, current-frequency ( $I$ - $f$ ) control, interior permanent magnet synchronous motor (IPMSM), sensorless control, startup strategy.

## I. INTRODUCTION

RECENTLY, the usage of permanent magnet synchronous motors (PMSMs) has increased in industrial applications due to their appealing features, such as high power density, high efficiency, robust rotor structure, and low maintenance cost [1]. In order to improve operation performance, PMSM requires

accurate and real-time rotor position, which is traditionally achieved by installing speed sensor, such as incremental encoder [2] or resolver [3]. However, these sensors reduce system reliability in harsh working conditions and increase overall cost and volume [4]. Therefore, robust sensorless control strategies are more preferred, which can be roughly classified into two categories [5]: the back electromotive-force (EMF) estimation methods [6]–[9] for medium-high speed applications and the signal injection methods [10]–[12] for medium-low speed applications. In order to achieve high performance over the entire speed range, the back EMF estimation method adapted to medium-high speed needs to be combined with a signal injection method for startup and low speed operation [13], which obviously complicates the overall control algorithm. In most applications, PMSMs do not operate at ultra-low speed except during startup. Therefore, the advantages of signal injection methods such as good dynamic response at low speed are insignificant compared to its disadvantages, such as complexity, loss, torque fluctuations and high noise. Desirably, a simple and effective startup strategy is more suitable than signal injection method to accelerate the motor to a speed where the back EMF is high enough for accurate estimation.

The  $V/f$  control widely used in induction motor is introduced into PMSMs and various improved  $V/f$  control methods have been proposed. Although the stability and efficiency of  $V/f$  control are improved by frequency correction [14] and amplitude correction [15], [16], the overall performance of open current loop  $V/f$  control is still far inferior than field-oriented control (FOC). Furthermore, due to different control structures and principles of operation, it is not always trouble free to transit from  $V/f$  control to FOC after initial startup.

Another simple and effective startup strategy is known as the current-frequency ( $I$ - $f$ ) control, which has closed-loop current regulator and thus solves the problem of uncontrolled current in  $V/f$  control. The startup process under  $I$ - $f$  control can be divided into three main stages including speed up, constant speed and transition to sensorless FOC [17]. Due to the low damping coefficient of PMSM itself, the damping ratio needs to be increased to improve system stability, which can be realized by frequency correction based on active power perturbations [18], [19]. The stability analysis and parameter design based on linearized motion model have been discussed in [20].

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During speed up stage, various frequency increasing modes of current vector are used, such as  $f \propto t$  [17],  $f \propto t^2$  [21], and  $f \propto (1 - \cos\omega t)$  [22]. However, there is no closed-loop frequency controller in these methods, which means the acceleration needs to be set relatively small to avoid losing synchronization during heavy-load startup. Actually, the acceleration should be adjusted with load. In [19], the acceleration is generated by a torque controller, which makes the output torque equal to a constant and large value.

After accelerating to the set speed, the current frequency remains constant while the current amplitude gradually decreases to improve efficiency and ensure a smooth transition to sensorless FOC. Similarly, the current amplitude should be adjusted with load. Using instantaneous reactive power feedback to adjust the current amplitude, the  $i_d = 0$  control for surface PMSM is realized in [20] and the maximum torque per ampere (MTPA) control for interior PMSM (IPMSM) is realized in [23]. Additionally, a current oriented I-f control with a closed-loop angle regulator design based on Lyapunov theory is proposed for MTPA realization [24].

Another key technique is the transition strategy from I-f startup to sensorless FOC. The smooth transition is achieved by a first-order lag compensator in [21]. A simple current cross-over technique is introduced, but is limited to light-load startup condition [25]. In [26], the output of position estimator and speed PI regulator are further processed by an adaptive compensator to generate suitable current amplitude and angle for smooth transition.

Generally speaking, the load capacity, startup speed and efficiency are the common requirements of I-f control, but the existing methods have shortcomings to achieve an optimal trade-off among them. Actually, these requirements can be naturally satisfied once the motor runs at MTPA state during three startup stages. However, lacking of accurate current angle information and corresponding angle control, the MTPA state under I-f control is particularly unstable. The stability can be enhanced by leaving some angle margin. For IPMSM, the current vector angle under MTPA state is slightly less than zero. Thus, sacrificing a little bit of efficiency, the  $i_d = 0$  operation, i.e., regulating current vector angle to zero, is adopted for higher stability margin, which achieves a good performance tradeoff. Hence, an angle controller is proposed in this article. Its input is the current vector angle, i.e., the angle error between current vector and  $q$ -axis, which can be easily obtained through current oriented voltage equation. During speed up stage, the current amplitude is set to the rated value for large startup torque while the current acceleration is the output of angle controller. Thus, the acceleration will be automatically adjusted with load to minimize angle error. Once the angle error is zero, the output torque is constant and large, which obviously shortens the acceleration period. After reaching the set speed, the current frequency remains constant while the current amplitude is the output of angle controller. Similarly, the angle controller will make angle error near zero by adjusting current amplitude with load. The core idea of the proposed angle controller is to maintain the angle error close to zero throughout the whole startup process, which ensures large startup torque, high efficiency and smooth transition from I-f

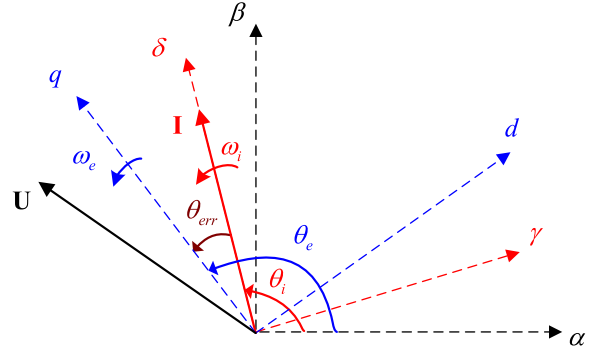


Fig. 1. Vector diagram including  $dq$  and  $\gamma\delta$  frames for IPMSM.

control to sensorless FOC. Moreover, this proposed I-f control has strong robustness because the angle error estimation is insensitive to parameter variations. So, this fast, efficient and robust I-f control has high practical value, even if the load is heavy during startup process.

## II. MATHEMATICAL IPMSM MODEL WITH I-F CONTROL

The voltage equations in the  $dq$  coordinate system are

$$u_d = R_s i_d + L_d \frac{d}{dt} i_d - \omega_e L_q i_q \quad (1)$$

$$u_q = R_s i_q + L_q \frac{d}{dt} i_q + \omega_e L_d i_d + \omega_e \lambda_f \quad (2)$$

where  $u_d$  and  $u_q$  represent  $d$ -axis and  $q$ -axis voltages, respectively,  $i_d$  and  $i_q$  denote  $d$ -axis and  $q$ -axis currents respectively,  $R_s$ ,  $L_d$ ,  $L_q$ ,  $\lambda_f$  and  $\omega_e$  represent stator resistance,  $d$ -axis inductance,  $q$ -axis inductance, permanent magnet flux linkage and electrical angular velocity respectively.

However, in practical applications, the I-f startup method is used in cases where the actual angle position is not known due to unreliable position signal or lack of a position sensor. Therefore, the actual  $dq$  coordinate frame is not known. In I-f control, the current vector is fully controllable. In order to analyze the characteristics of current vector, the  $\gamma\delta$  coordinate system oriented by the current vector is introduced in Fig. 1.

The coordinate systems  $\alpha\beta$  and  $dq$  are stationary and synchronous frames, respectively. The  $\delta$ -axis is aligned with the current vector  $\mathbf{I}$ , the angle between current vector  $\mathbf{I}$  and  $\alpha$  axis is denoted by  $\theta_i$ , and the angle between  $q$ -axis and  $\alpha$ -axis is denoted by  $\theta_e$ . The angle between current vector  $\mathbf{I}$  and  $q$  axis is  $\theta_{err}$ , satisfying  $\theta_{err} = \theta_e - \theta_i$ . The frequencies  $\omega_i$  and  $\omega_e$  represent current vector and motor angular velocities respectively.

Since the angle error between  $q$ -axis and  $\delta$ -axis is  $\theta_{err}$ , the voltage equations in the  $\gamma\delta$  coordinate system are

$$u_\gamma = R_s i_\gamma + L_d \frac{d}{dt} i_\gamma - \omega_i L_q i_\delta - \omega_e \lambda_f \sin\theta_{err} \quad (3)$$

$$u_\delta = R_s i_\delta + L_q \frac{d}{dt} i_\delta + \omega_i L_d i_\gamma + \omega_e \lambda_f \cos\theta_{err} \quad (4)$$

where  $u_\gamma$  and  $u_\delta$  represent  $\gamma$ -axis and  $\delta$ -axis voltages, respectively,  $i_\gamma$  and  $i_\delta$  denote  $\gamma$ -axis and  $\delta$ -axis currents, respectively.

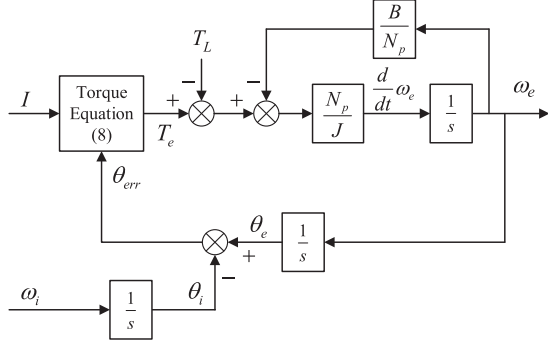


Fig. 2. Large-signal motion model of IPMSM under  $I$ - $f$  control.

Since  $\delta$ -axis is aligned with the current vector  $\mathbf{I}$ , the currents satisfy  $i_\gamma = 0$ ,  $i_\delta = I_m$ , where  $I_m$  is the amplitude of current vector  $\mathbf{I}$ . Then the  $\gamma\delta$ -axis voltage equations can be simplified as [23]

$$u_\gamma = -\omega_i L_q i_\delta - \omega_e \lambda_f \sin\theta_{err} \quad (5)$$

$$u_\delta = R_s i_\delta + L_q \frac{d}{dt} i_\delta + \omega_e \lambda_f \cos\theta_{err}. \quad (6)$$

According to the relationship between  $dq$  and  $\gamma\delta$  coordinate systems, the derivative of angle error can be obtained as

$$\frac{d}{dt} \theta_{err} = \frac{d}{dt} (\theta_e - \theta_i) = \omega_e - \omega_i. \quad (7)$$

The torque equation of IPMSM is

$$\begin{aligned} T_e &= 1.5 N_p i_q [\lambda_f + (L_d - L_q) i_d] \\ &= 1.5 N_p I_m \cos\theta_{err} [\lambda_f + (L_d - L_q) I_m \sin\theta_{err}] \end{aligned} \quad (8)$$

where the currents satisfy  $i_q = I_m \cos\theta_{err}$ ,  $i_d = I_m \sin\theta_{err}$ , and  $N_p$  is number of pole pairs.

The motor motion equation is

$$\frac{J}{N_p} \frac{d}{dt} \omega_e = T_e - T_L - \frac{B}{N_p} \omega_e \quad (9)$$

where  $J$  is moment of inertial,  $T_L$  is load torque and  $B$  is damping coefficient.

Since the current response is fast enough compared to the motion response, the actual current vector can be regarded the same as the given current vector of current regulator when considering the motion state. Then, combined with motion (9), torque (8) and angle error (7), the large-signal motion model of IPMSM under  $I$ - $f$  control can be obtained as shown in Fig. 2.

### III. PERFORMANCE ANALYSIS OF CONVENTIONAL $I$ - $F$ STARTUP METHOD

The conventional  $I$ - $f$  startup method can be divided into three stages. During the speed up stage, the current vector amplitude is maintained constant as rated value while the current vector frequency  $\omega_i$  increases with time. When the speed reaches the set value, the motor switches to constant speed stage, where the current vector frequency is maintained constant and the current amplitude decreases gradually. As the current amplitude decreases, the angle error  $\theta_{err}$  between current vector and  $q$ -axis

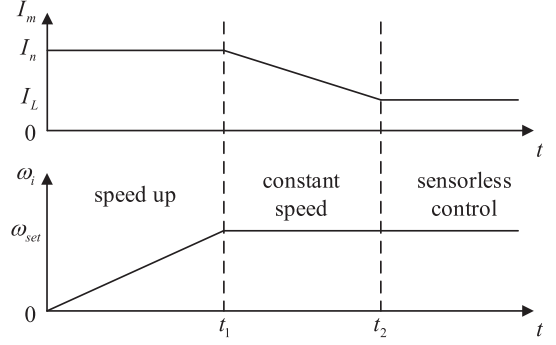


Fig. 3. Current vector amplitude and speed of conventional  $I$ - $f$  control.

TABLE I  
TESTED IPMSM PARAMETERS

Parameter	Value	Parameter	Value
Rated power $P_n$	1.5 kW	Resistance $R_s$	4.8 $\Omega$
Rated voltage $V_n$	380 Vrms	$d$ -axis inductance $L_d$	31.5 mH
Rated current $I_n$	2.7 Arms	$q$ -axis inductance $L_q$	92.3 mH
Rated speed $n$	1500 rpm	Pole pairs $N_p$	3
Rated torque $T_n$	9.55 Nm	Flux linkage $\lambda_f$	0.67 Wb
Inertia $J$	0.019 kgm <sup>2</sup>	Damping coefficient $B$	0.015 Nms

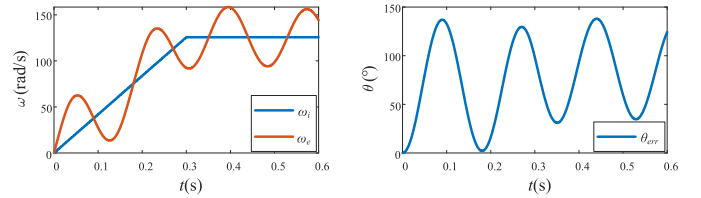


Fig. 4. Speed and angle error under conventional  $I$ - $f$  control at no load.

decreases as well. When the angle error  $\theta_{err}$  decreases to a relatively small value, the strategy switches to sensorless FOC. The current vector amplitude and frequency of conventional  $I$ - $f$  control during the entire startup sequence are shown in Fig. 3.

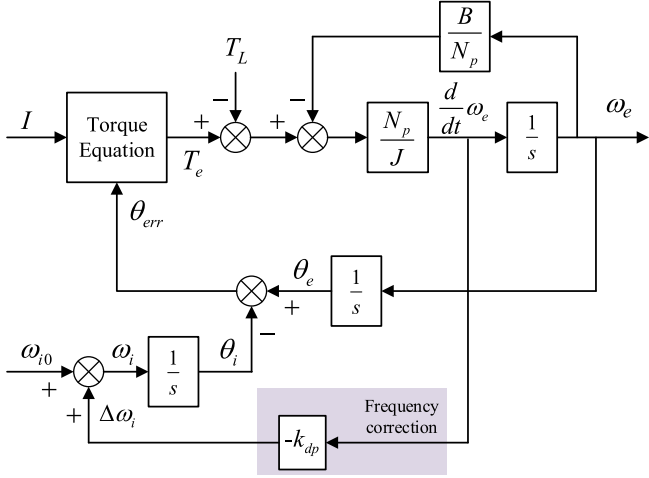
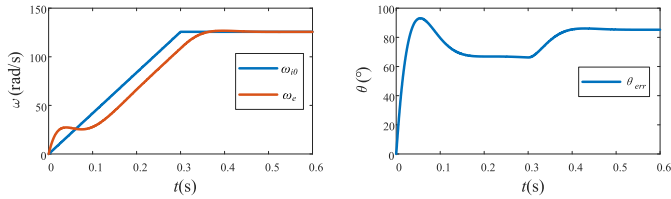
However, there are problems in the conventional  $I$ - $f$  startup strategy, such as speed oscillation caused by low damping ratio and losing synchronization during speed up stage. This section mainly analyzes these two issues.

#### A. Speed Oscillation Caused By Low Damping Ratio

The motor parameters are given in Table I, where the damping coefficient  $B$  is relatively small. After applying the current vector in Fig. 3 to the large-signal motion model in Fig. 2, the speed and angle error under  $I$ - $f$  control at no load are recorded in Fig. 4. Clearly, in this case, the speed and angle errors oscillate in a great deal. The detailed stability analysis can be referred to [20].

In order to solve the speed oscillation caused by low damping ratio, a frequency correction method based on active power fluctuation is proposed in [18]. The correction frequency  $\Delta\omega_i$  of current vector is proportional to the derivative of motor operating frequency  $\frac{d}{dt} \omega_e$ , i.e.,

$$\Delta\omega_i = \omega_i - \omega_{i0} = -k_{dp} \frac{d}{dt} \omega_e \quad (10)$$


 Fig. 5. Large-signal motion model under  $I$ - $f$  control with frequency correction.

 Fig. 6. Speed and angle error under  $I$ - $f$  control with frequency correction at no load.

where  $\omega_{i0}$  and  $\omega_i$  are current vector frequency without and with frequency correction respectively. The gain  $k_{dp}$  for damping is a positive constant. The derivative of motor frequency can be obtained through high-pass filtering of the active power. The specific derivation process and parameter selection will be described in detail in the following section.

With frequency correction (10), the large-signal motion model under  $I$ - $f$  control can be obtained as shown in Fig. 5, and the speed and angle error at no load are recorded in Fig. 6. In this case, there are no speed oscillation and the motor frequency  $\omega_e$  lags behind the set current vector frequency  $\omega_i$ . This is because the system equivalent damping ratio is increased by the frequency correction.

### B. Losing Synchronization During Speed Up Stage

Under conventional  $I$ - $f$  control, the motor often loses synchronization especially during high-load or high-acceleration startup. This is because the output torque cannot reach the demand torque, which needs to be quantitatively analyzed. According to the motion (9), the demand torque with load  $T_L$  and acceleration  $\beta = \frac{d}{dt} \omega_e$  satisfies

$$T_d = T_L + \frac{B}{N_p} \omega_e + \frac{J}{N_p} \beta. \quad (11)$$

Obviously, the larger the load  $T_L$  or acceleration  $\beta$  is, the larger the demand torque  $T_d$  is.

According to torque (8), the motor output torque  $T_e$  with respect to angle error  $\theta_{err}$  is recorded in Fig. 7. When the motor

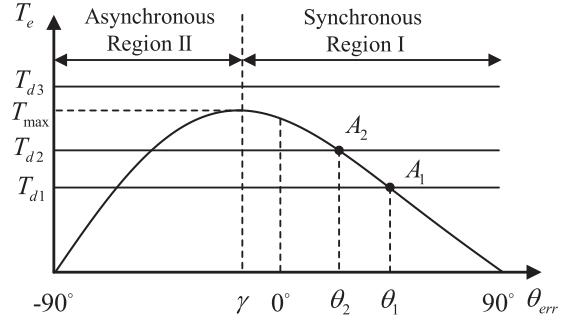


Fig. 7. Variation of generated torque with angle error.

runs at MTPA state, the output torque  $T_e$  reaches its maximum value  $T_{max}$ , and the present angle error  $\theta_{err}$  is equal to the MTPA angle  $\gamma$ , which satisfies

$$\gamma = \sin^{-1} \frac{\lambda_f - \sqrt{\lambda_f^2 + 8(L_d - L_q)^2 I^2}}{4(L_q - L_d)I} < 0. \quad (12)$$

Based on whether the angle error  $\theta_{err}$  is greater than the MTPA angle  $\gamma$ , the operating region can be divided into synchronous region I and asynchronous region II as shown in Fig. 7. When load  $T_L$  increases suddenly, the angle error  $\theta_{err}$  will decrease according to (9). Then, in synchronous region I, as the angle error  $\theta_{err}$  decreases, the output torque  $T_e$  will increase. When it increases to the demand torque  $T_d$ , the motor will be stable again. However, in asynchronous region II, as the angle error  $\theta_{err}$  decreases, the output torque  $T_e$  will decrease, and thus the angle error  $\theta_{err}$  will further decrease according to (9), which makes motor lose synchronization. Therefore, it is necessary to ensure the motor runs in synchronous region I. Furthermore, when the motor runs in region I, if the demand torque increases from  $T_{d1}$  to  $T_{d2}$ , the steady-state operating point will switch from point  $A_1$  to  $A_2$  accordingly as shown in Fig. 7. However, when the demand torque increases to  $T_{d3}$ , which is larger than the maximum output torque  $T_{max}$ , the operating point keeps moving to the left. Once it enters asynchronous region II, the motor will lose synchronization.

With the same acceleration in Fig. 6 and at full load, the speed, angle error and torque are recorded in Fig. 8. When the demand torque  $T_d$  is less than maximum torque  $T_{max}$ , the steady-state operating point is always in synchronous region I, where the output torque  $T_e$  gradually approaches the demand torque  $T_d$ . However, according to (11), when the load  $T_L$  and acceleration  $\beta$  are large, as the speed  $\omega_e$  increases, the demand torque  $T_d$  may be larger than the maximum torque  $T_{max}$ , leading to the loss of synchronization. It can be seen more intuitively from the angle error waveform that once the angle error  $\theta_{err}$  is smaller than MTPA angle  $\gamma$ , the operating point enters asynchronous region II and thus the angle error  $\theta_{err}$  keeps on decreasing, resulting in loss of synchronization and startup failure.

Therefore, the conventional  $I$ - $f$  startup method suffers from losing synchronization as described and thus an improved method is needed to refine the  $I$ - $f$  control.

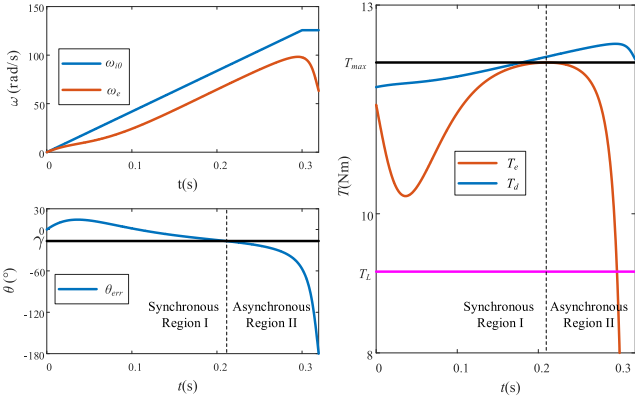


Fig. 8. Speed, angle error and torque under  $I$ - $f$  control with frequency correction at full load.

#### IV. PROPOSED $I$ - $f$ STARTUP METHOD BASED ON ANGLE CONTROLLER

From previous analysis, it is clear that the motor can lose synchronization when the demand torque  $T_d$  is greater than the maximum output torque  $T_{\max}$  during speed up stage. According to (11), the demand torque  $T_d$  mainly consists of the load torque  $T_L$  and the acceleration torque  $\frac{J}{N_p}\beta$  for speeding up with acceleration  $\beta$ . Since the load torque is determined by external working condition, the controllable torque is the acceleration torque. Therefore, the demand torque  $T_d$  can be controlled by adjusting acceleration  $\beta$  to be smaller than the maximum output torque  $T_{\max}$ . In other words, when the load torque  $T_L$  is large, the demand torque  $T_d$  can be avoided from exceeding  $T_{\max}$  by decreasing acceleration  $\beta$ . In fact, it follows from Fig. 7 that maintaining  $T_d < T_{\max}$  is essential to keep  $\theta_{err} > \gamma$ . Therefore, the demand torque  $T_d$  can be ensured smaller than the maximum torque  $T_{\max}$  by controlling  $\theta_{err} = 0$  to avoid the loss of synchronization. When  $\theta_{err} = 0$ , the output startup torque is relatively large and close to its maximum value  $T_{\max}$ .

When the motor accelerates to the set speed and switches to constant speed stage, the acceleration  $\beta$  is zero, causing demand torque  $T_d$  to decrease sharply according to (11). If the motor current remains unchanged, the angle error  $\theta_{err}$  will increase significantly as demand torque  $T_d$  decreases according to Fig. 7, which reduces the operating efficiency. In order to improve efficiency, it is necessary to reduce motor current and  $T_{\max}$ , thus the angle error  $\theta_{err}$  will decrease. Similarly, the angle error  $\theta_{err}$  needs to be larger than the MTPA angle  $\gamma$  to avoid losing synchronization. Therefore, the motor current can be adjusted to maintain  $\theta_{err} = 0$ , which not only realizes high efficiency, but also facilitates subsequent smooth transition to sensorless FOC on the premise of avoiding losing synchronization.

Combined with the previous analysis, an improved  $I$ - $f$  startup method based on angle controller is proposed as shown in Fig. 9. The core idea of the proposed method is to adjust the acceleration of current vector during speed up stage and adjust the amplitude of current vector during constant speed stage to maintain angle

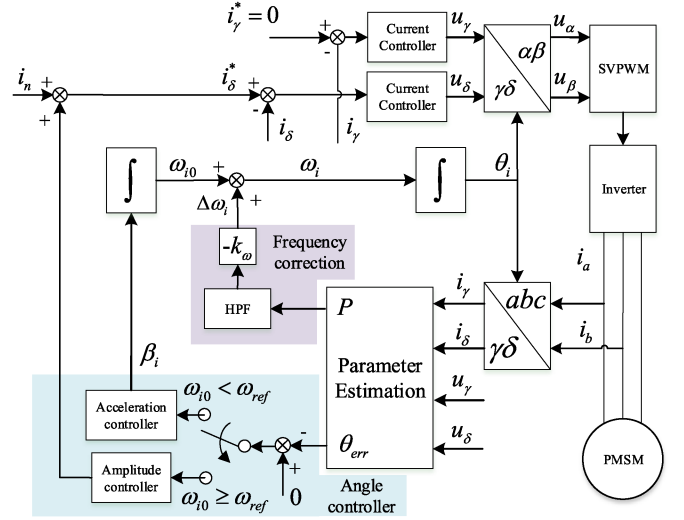


Fig. 9. Proposed  $I$ - $f$  startup method with angle controller.

error  $\theta_{err} = 0$ . Thus, throughout the whole  $I$ - $f$  startup process, the angle error  $\theta_{err}$  is guaranteed to be zero, which ensures large startup torque, high efficiency and smooth transition to subsequent sensorless FOC.

Since the torque equation in large-signal motion model shown in Fig. 2 is nonlinear, the performance analysis and controller design are both complicated. Therefore, by linearizing the torque (8), the small-signal model of torque equation can be obtained as

$$\Delta T_e = K_I \cdot \Delta I - K_\theta \cdot \Delta \theta_{err} \quad (13)$$

where  $K_I$  and  $K_\theta$  are small-signal gains, satisfying

$$K_I = \frac{\partial T_e}{\partial I} = \frac{3}{2} N_p [(L_d - L_q) I \sin 2\theta_{err} + \lambda_f \cos \theta_{err}] \quad (14)$$

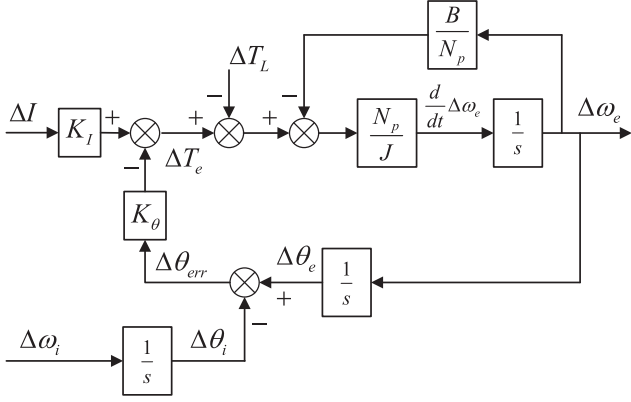
$$K_\theta = -\frac{\partial T_e}{\partial \theta_{err}} = \frac{3}{2} N_p [(L_q - L_d) I^2 \cos 2\theta_{err} + \lambda_f I \cos \theta_{err}]. \quad (15)$$

Very often, in order to guarantee a very small initial  $\theta_{err}$ , the rotor position initialization is carried out before  $I$ - $f$  startup, such as applying a fixed current vector to align the rotor [20] or detecting the initial position through high frequency pulsating voltage injection [27]. After startup, the proposed angle controller works to bring the angle error  $\theta_{err}$  to zero. Therefore, the steady-state angle error  $\theta_{err}$  is close to zero during  $I$ - $f$  startup. So the small-signal gains (14) and (15) can be simplified as

$$K_I = \frac{\partial T_e}{\partial I} \Big|_{\theta_{err}=0} = \frac{3}{2} N_p \lambda_f \quad (16)$$

$$K_\theta = -\frac{\partial T_e}{\partial \theta_{err}} \Big|_{\theta_{err}=0} = \frac{3}{2} N_p (L_q - L_d) I^2. \quad (17)$$

Combined with small-signal torque (13), the small-signal motion model in Fig. 10 can be derived from large-signal motion model shown in Fig. 2.

Fig. 10. Small-signal motion model of IPMSM under  $I$ - $f$  control.

The small-signal angle error (18) and motion (19) as below are derived from (7) and (9), respectively,

$$\Delta\theta_{\text{err}} = \frac{1}{s}(\Delta\omega_e - \Delta\omega_i) \quad (18)$$

$$\Delta\omega_e = \frac{N_p}{sJ + B}(\Delta T_e - \Delta T_L). \quad (19)$$

Combing (13), (18), and (19), the relationship among small-signal angle error  $\Delta\theta_{\text{err}}$ , current amplitude  $\Delta I$ , current frequency  $\Delta\omega_i$  and load torque  $\Delta T$  can be derived as

$$\Delta\theta_{\text{err}} = G_{\theta I}\Delta I + G_{\theta\omega}\Delta\omega_i + G_{\theta T}\Delta T \quad (20)$$

where the transfer functions satisfy

$$G_{\theta I} = \frac{N_p K_I}{s^2 J + sB + N_p K_\theta} \quad (21)$$

$$G_{\theta\omega} = -\frac{sJ + B}{s^2 J + sB + N_p K_\theta} \quad (22)$$

$$G_{\theta T} = -\frac{N_p}{s^2 J + sB + N_p K_\theta}. \quad (23)$$

Based on the above transfer functions, the sections below analyze how to control the current vector amplitude  $I$  and frequency  $\omega_i$  to make the angle error  $\theta_{\text{err}}$  converge to zero stably and quickly.

#### A. Frequency Correction Design

According to transfer functions (21)–(23), the characteristic equation of the system under  $I$ - $f$  control is

$$s^2 J + sB + N_p K_\theta = 0. \quad (24)$$

The damping ratio  $\zeta$  of the characteristic (24) is

$$\zeta = \frac{B}{2\sqrt{JN_p K_\theta}} \quad (25)$$

According to the motor parameters in Table I, it can be seen that the damping coefficient  $B$  is relatively small, so the system damping ratio  $\zeta$  is small, leading to large speed oscillation during startup stage as shown in Fig. 4. In order to solve the speed oscillation caused by low damping ratio, a frequency correction method is proposed in [19]. The equation of correction frequency

$\Delta\omega_i$  satisfies (10) and the corresponding small-signal equation is expressed as

$$\Delta\omega_i = \Delta\omega_{i0} - k_{dp} \frac{d}{dt} \Delta\omega_e \quad (26)$$

where  $\Delta\omega_{i0}$  and  $\Delta\omega_i$  are current vector frequency perturbations without and with frequency correction respectively, and  $\Delta\omega_e$  is motor frequency perturbation. Substituting (26) into (18), the angle error  $\theta_{\text{err}}$  with frequency correction can be obtained as

$$\Delta\theta_{\text{err}} = \frac{1}{s} [(1 + sk_{dp}) \Delta\omega_e - \Delta\omega_{i0}]. \quad (27)$$

Combining (13), (19), and (27), the transfer functions with frequency correction are derived as

$$G'_{\theta I} = \frac{N_p K_I (sk_{dp} + 1)}{s^2 J + sB_e + N_p K_\theta} \quad (28)$$

$$G'_{\theta\omega} = -\frac{sJ + B}{s^2 J + sB_e + N_p K_\theta} \quad (29)$$

$$G'_{\theta T} = -\frac{N_p (sk_{dp} + 1)}{s^2 J + sB_e + N_p K_\theta} \quad (30)$$

where  $B_e$  is the equivalent damping coefficient with frequency correction and satisfies

$$B_e = B + N_p K_\theta k_{dp} \approx N_p K_\theta k_{dp} \quad (B \ll N_p K_\theta k_{dp}). \quad (31)$$

Comparing the transfer functions (21)–(23) and (28)–(30), it's noteworthy that with frequency correction, the equivalent damping coefficient increases from  $B$  to  $B_e$ . To obtain faster convergence with less oscillation, the equivalent damping ratio  $\zeta_e$  with frequency correction is set to  $1/\sqrt{2}$  here, i.e.,

$$\zeta_e = \frac{B_e}{2\sqrt{JN_p K_\theta}} = \frac{1}{\sqrt{2}}. \quad (32)$$

Substituting (31) into (32), the frequency correction gain  $k_{dp}$  is expressed as

$$k_{dp} = \sqrt{\frac{2J}{N_p K_\theta}}. \quad (33)$$

Furthermore, it is necessary to obtain the derivative of motor frequency. According to active power equation, the motor frequency satisfies  $\omega_e = P/T_e$ . Since the angle error is kept to be zero and the current amplitude  $I_m$  is set to its rated value  $I_n$ , the torque  $T_e$  is nearly constant during startup. Therefore, the derivative of motor frequency  $d\omega_e/dt$  is proportional to the derivative of active power  $dP/dt$ . According to Fig. 4, the motor oscillation frequency is very low, only several hertz. In low frequency, high-pass filter (HPF) is almost a differentiator. Therefore, the derivative of motor frequency can be obtained by filtering active power  $P$  through HPF and the equation is expressed as

$$\frac{d}{dt} \omega_e = \frac{d}{dt} \left( \frac{P}{T_e} \right) \approx \frac{\text{HPF}(P)}{T_e}. \quad (34)$$

The active power  $P$  and output torque  $T_e$  can be estimated by the following equations:

$$\begin{aligned} P &\approx [u_a \ u_b \ u_c] \begin{bmatrix} i_a \\ i_b \\ i_c \end{bmatrix} = \left( \mathbf{M}_{\alpha\beta 2abc} \begin{bmatrix} u_\alpha \\ u_\beta \end{bmatrix} \right)^T \left( \mathbf{M}_{\alpha\beta 2abc} \begin{bmatrix} i_\alpha \\ i_\beta \end{bmatrix} \right) \\ &= [u_\alpha \ u_\beta] \left( \mathbf{M}_{\alpha\beta 2abc}^T \mathbf{M}_{\alpha\beta 2abc} \right) \begin{bmatrix} i_\alpha \\ i_\beta \end{bmatrix} = 1.5 [u_\alpha \ u_\beta] \begin{bmatrix} i_\alpha \\ i_\beta \end{bmatrix} \end{aligned} \quad (35)$$

$$T_e = 1.5 N_p \lambda_f I_n \quad (36)$$

where the  $\mathbf{M}_{\alpha\beta 2abc}$  is the inverse Clark transform matrix, satisfying  $\mathbf{M}_{\alpha\beta 2abc} = \begin{bmatrix} 1 & -1/2 & -1/2 \\ 0 & \sqrt{3}/2 & -\sqrt{3}/2 \end{bmatrix}^T$ .

Therefore, the active power perturbation gain  $k_\omega$  in Fig. 9 satisfies

$$k_\omega = \frac{k_{dp}}{T_e} = \frac{2}{3 N_p \lambda_f I} \sqrt{\frac{2J}{N_p K_\theta}}. \quad (37)$$

### B. Acceleration Controller Design

During speed up stage, i.e., when current vector frequency  $\omega_{i0}$  is less than set frequency  $\omega_{ref}$ , the current vector amplitude  $I$  is set to rated value  $I_n$  to ensure a relatively large startup torque. According to previous analysis, when the load torque  $T_L$  is large, the current vector acceleration  $\beta_i$  should be smaller to avoid losing synchronization. When the load torque  $T_L$  is small, the current vector acceleration  $\beta_i$  should be larger to shorten the startup time. Therefore, the acceleration  $\beta_i$  should be automatically adjusted as a function of the load torque  $T_L$ . With the proposed angle controller, the angle error can be maintained as zero, and the motor frequency  $\omega_e$  is close to the current vector frequency  $\omega_i$ . Combining torque (8) and motion (9), the relationship between load torque  $T_L$  and current vector acceleration  $\beta_i$  satisfies

$$T_e = \frac{3}{2} N_p \lambda_f I_n = T_L + \frac{J}{N_p} \beta_i + \frac{B}{N_p} \omega_i. \quad (38)$$

It is clear that when the output torque is constant and relatively large, the acceleration  $\beta_i$  is able to adjust automatically with the load torque  $T_L$  and is relatively large on the premise of avoiding losing synchronization. Therefore, during speed up stage, maintaining zero angle error  $\theta_{err}$  ensures large startup torque, short startup time and avoids losing synchronization.

According to  $\gamma$ -axis voltage (5), the angle error  $\theta_{err}$  can be estimated as

$$\theta_{err} \approx \sin \theta_{err} = \frac{-\omega_i L_q i_\delta - u_\gamma}{\omega_e \lambda_f} \approx \frac{-\omega_i L_q i_\delta - u_\gamma}{\omega_i \lambda_f}. \quad (39)$$

The angle error  $\theta_{err}$  is assumed as  $\sin \theta_{err}$  because it is small due to rotor position initialization and zero angle error controller as analyzed before. The motor frequency  $\omega_e$  can be approximated as the current vector frequency  $\omega_i$ . Actually, the speed error  $\omega_e - \omega_i$  determines the derivative of angle error  $\theta_{err}$  according to (7). Since the angle error  $\theta_{err}$  fluctuates very little

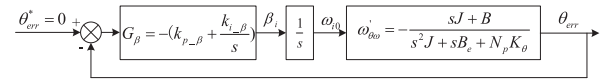


Fig. 11. Small-signal current acceleration closed loop control diagram.

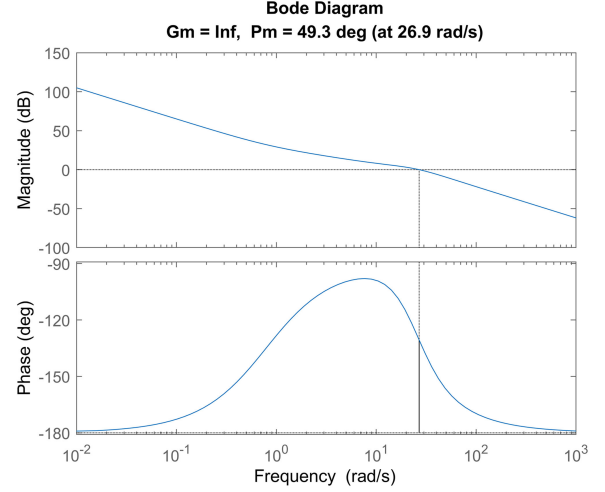


Fig. 12. Bode diagram of current acceleration open loop transfer function.

after frequency correction as shown in Fig. 6, the speed error is relatively small.

After obtaining angle error, the acceleration controller adjusts the current vector acceleration  $\beta_i$ . The current vector frequency  $\omega_{i0}$  obtained from the integration of acceleration  $\beta_i$  affects the angle error  $\theta_{err}$  through transfer function  $G'_{\theta\omega}$  to make the angle error  $\theta_{err}$  near zero. The small-signal current acceleration closed loop control diagram is shown in Fig. 11. The acceleration controller here uses a robust PI controller to track the reference value  $\theta_{err}^*$  without steady-state error.

The open loop transfer function  $G_{open\_beta}$  of acceleration control system is

$$G_{open\_beta} = \frac{1}{s} G_\beta G'_{\theta\omega} = \frac{1}{s} \left( k_p - \beta + \frac{k_i}{s} \right) \frac{sJ + B}{s^2 J + sB_e + N_p K_\theta}. \quad (40)$$

Considering that the initial angle error  $\theta_{err}$  at the starting moment may not be close to zero, the acceleration control loop bandwidth should be set smaller to ensure normal startup. In this case, it is set to 4 Hz, which is 1% of the bandwidth of current control loop and the phase margin is about  $50^\circ$ . Fig. 12 shows the Bode diagram of current acceleration open loop transfer function  $G_{open\_beta}$ .

### C. Amplitude Controller Design

During speed up stage, once the current vector frequency  $\omega_{i0}$  is greater than the set frequency  $\omega_{ref}$ , the current vector frequency  $\omega_{i0}$  is maintained as the set value  $\omega_{ref}$  and thus the motor switches to constant speed stage. As analyzed before, in order to improve efficiency, the motor current needs to be reduced to maintain zero angle error  $\theta_{err}$ .



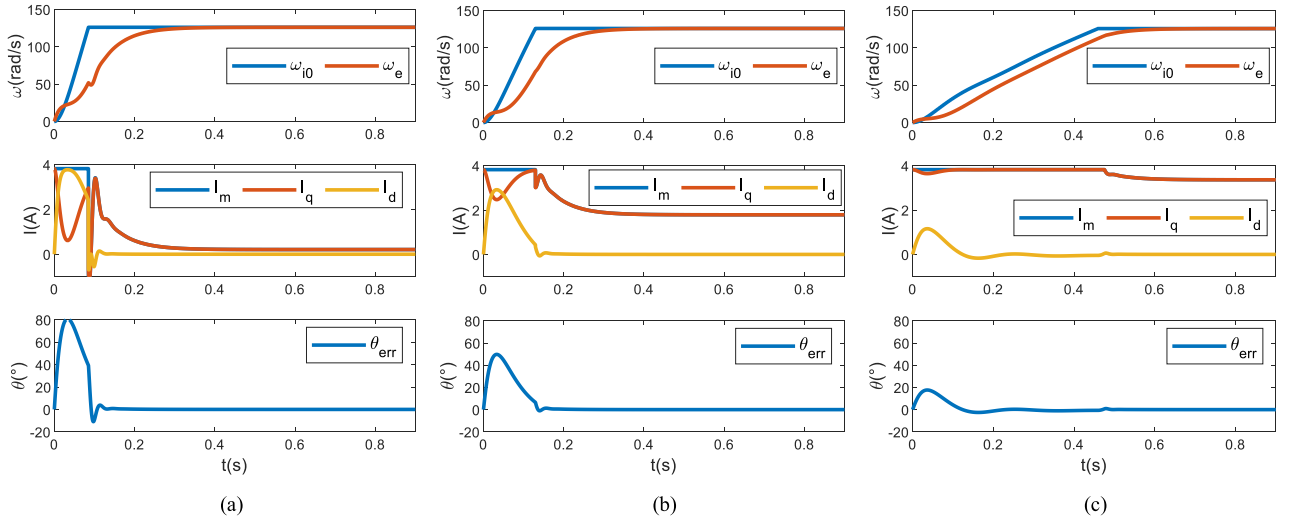


Fig. 16. Simulation results under  $I$ - $f$  control with angle controller at different loads. From top to bottom: speed, current amplitude, angle error.

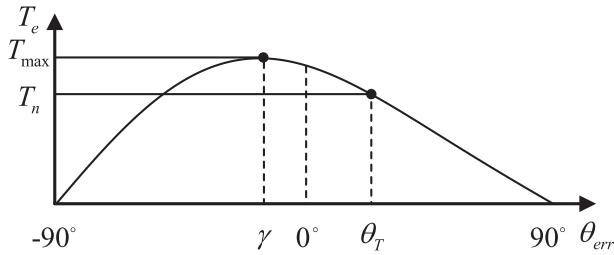


Fig. 17. Variation of generated torque with angle error at rated current.

multiplier gain of  $\frac{\theta_{\text{err}}}{\theta_{\text{err}}^e} = \frac{\lambda_f^e}{\lambda_f}$ . As long as the angle controller is reasonably designed, i.e., the system phase margin is relatively large, the system can work properly. It means that the impact of inaccurate estimated flux value  $\lambda_f^e$  is relatively small.

When the estimated flux value  $\lambda_f^e$  is equal to its actual value  $\lambda_f$ , (43) can be simplified as

$$\theta_{\text{err}} = \theta_{\text{err}}^e + (L_q^e - L_q) \frac{i_\delta}{\lambda_f}. \quad (45)$$

In this case, there is a dc bias between actual angle error  $\theta_{\text{err}}$  and its estimated value  $\theta_{\text{err}}^e$ . Assuming that the angle controller can work properly, the estimated angle error  $\theta_{\text{err}}^e$  will be near zero, so the actual steady-state angle error  $\theta_{\text{err}}$  satisfies

$$\theta_{\text{err}} = (L_q^e - L_q) \frac{i_\delta}{\lambda_f}. \quad (46)$$

Obviously, the larger the current vector amplitude  $i_\delta$  is, the larger the absolute value of angle error  $|\theta_{\text{err}}|$  will be. Therefore, considering the case with the largest absolute value of angle error, i.e., when the current vector amplitude  $i_\delta$  is set to the rated current value  $I_n$ , the variation of output torque with angle error is recorded in Fig. 17.

In order to avoid losing synchronization, as previously analyzed, the angle error  $\theta_{\text{err}}$  needs to be greater than MTPA angle  $\gamma$ . In addition, the output torque  $T_e$  must be at least greater than the rated load  $T_n$  to ensure smooth startup under maximum load. When the output torque  $T_e$  equals to the rated load  $T_n$ , the angle error is marked as load angle  $\theta_T$ , which satisfies

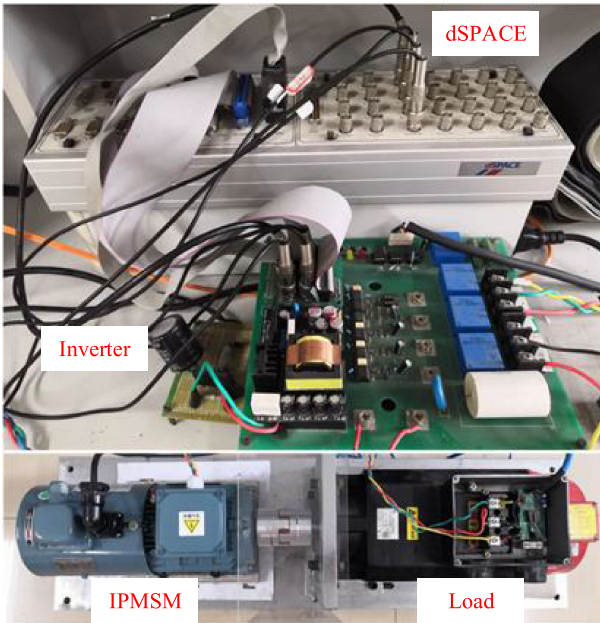
$$T_n = 1.5N_p I_n \cos\theta_T [\lambda_f + (L_d - L_q) I_n \sin\theta_T], \quad \theta_T > 0. \quad (47)$$

So, the actual angle error  $\theta_{\text{err}}$  needs to meet

$$\gamma < \theta_{\text{err}} < \theta_T. \quad (48)$$

Fig. 18. Experimental platform.

In this case, the actual angle error  $\theta_{\text{err}}$  is proportional to its estimated value  $\theta_{\text{err}}^e$ . Refer to Fig. 9, the estimated input error of angle controller is proportional to its actual value with a



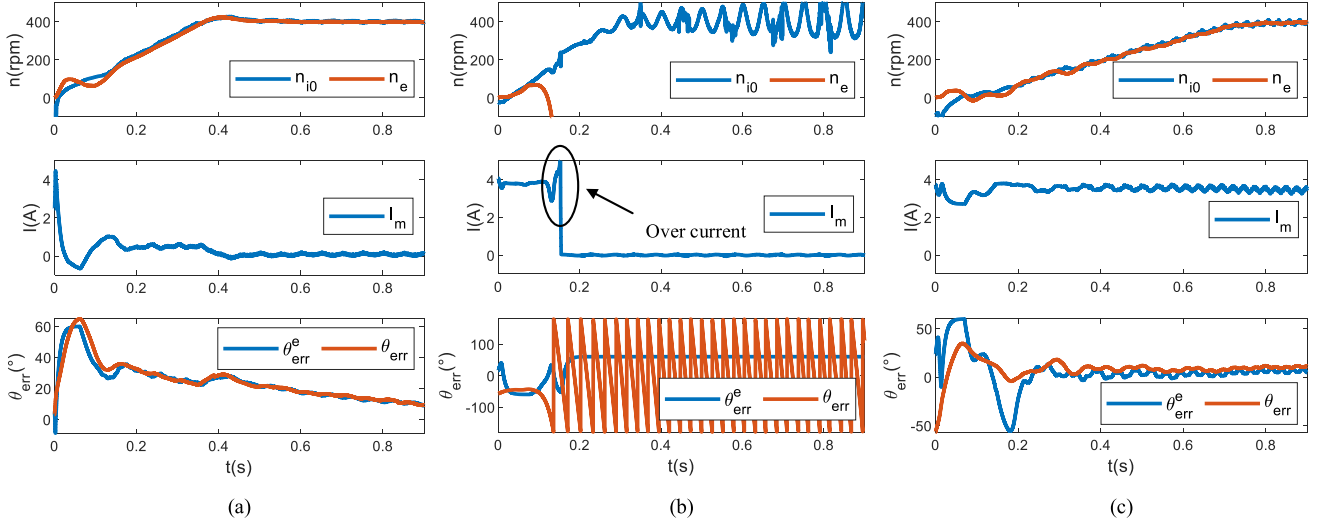


Fig. 19. Experimental results under existing  $I$ - $f$  control with frequency correction and amplitude controller at different conditions. From top to bottom: speed; current amplitude; angle error. (a) High acceleration and no load. (b) High acceleration and full load. (c) Low acceleration and full load.

Substituting (46) into (48), the range of estimated inductance  $L_q^e$  can be obtained as

$$1 + \frac{\lambda_f}{L_q I_n} \gamma < \frac{L_q^e}{L_q} < 1 + \frac{\lambda_f}{L_q I_n} \theta_T. \quad (49)$$

Combined with the motor parameters in Table I, the estimated inductance value at rated current should satisfy

$$44.3\% < \frac{L_q^e}{L_q} < 168\%. \quad (50)$$

It can be seen that the estimated inductance value  $L_q^e$  can vary over a wide range without any performance degradation.

When the estimated flux and inductance are different than their actual values, the actual angle error in (43) can be rearranged as follows:

$$\theta_{err} = \frac{\lambda_f^e}{\lambda_f} \left[ \theta_{err}^e + \frac{\lambda_f}{\lambda_f^e} (L_q^e - L_q) \frac{i_\delta}{\lambda_f} \right]. \quad (51)$$

In fact, the actual angle error is linear to its estimated value, which is a synthesis of the above two cases. Similarly, the effect of gain  $\frac{\lambda_f^e}{\lambda_f}$  can be neglected due to the robustness of PI controller, and the motor is still stable as long as the dc bias  $\frac{\lambda_f}{\lambda_f^e} (L_q^e - L_q) \frac{i_\delta}{\lambda_f}$  is within a certain range, i.e., greater than MTPA angle  $\gamma$  and less than load angle  $\theta_T$ .

In summary, the proposed  $I$ - $f$  startup strategy with current angle controller has a strong robustness performance.

## VI. EXPERIMENTAL RESULTS

In order to validate the proposed strategy, the experiments are carried out on the test platform as shown in Fig. 18, including dSPACE controller, inverter, load motor and the tested IPMSM. The parameters of this IPMSM are given in Table I. The switching frequency and control frequency are both at 4 kHz.

### A. Performance of Existing $I$ - $f$ Startup Strategy

The existing  $I$ - $f$  startup strategy [28] adopts frequency correction and amplitude controller with a preset and constant acceleration during speed up stage. It is noteworthy that the amplitude controller is used in speed up stage for higher efficiency and used in constant speed stage for subsequent transition to FOC. The experimental waveforms under high acceleration and no load condition are recorded in Fig. 19(a). After implementing the frequency correction algorithm, there is basically no oscillation during startup. Because of the amplitude controller, the current amplitude  $I_m$  and angle error  $\theta_{err}$  are relatively small throughout the whole process. However, when the motor starts at full load with faster acceleration, the motor loses synchronization and the results are recorded in Fig. 19(b). According to (11), the demand torque  $T_d$  is very large under high acceleration and high load condition. Once the demand torque  $T_d$  is greater than the maximum output torque  $T_{max}$ , the motor will be out of step and the motor current will increase dramatically until the overcurrent protection is triggered. Using this strategy, in order to start smoothly even at full load, the acceleration is usually set to a very small value to meet  $T_d < T_{max}$ , and the corresponding results are recorded in Fig. 19(c). Therefore, the existing  $I$ - $f$  startup strategy needs to set a slower acceleration to ensure smooth startup at full load, which degrades the startup speed performance. In this case, the acceleration is set to be a constant and low value, and it cannot be adjusted automatically with the actual load, which means the acceleration is very small even under light load for successful startup.

### B. Performance of Proposed $I$ - $f$ Startup Strategy

To demonstrate large startup torque, high efficiency capability of the proposed  $I$ - $f$  startup strategy with current angle controller, the experimental results under different load conditions are recorded in Fig. 20. Regardless of at no load, half load or full load, the motor starts smoothly. Moreover, the proposed method

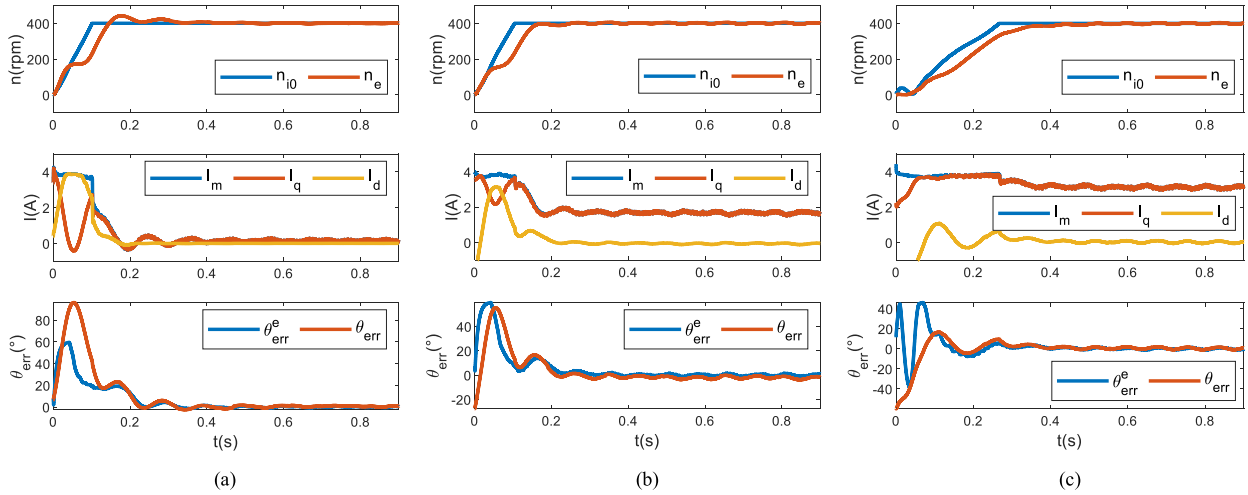


Fig. 20. Experimental results under proposed  $I$ - $f$  control with angle controller at different loads. From top to bottom: speed; current amplitude; and angle error. (a) No load. (b) Half rated load. (c) Rated load.

does not need to set the startup acceleration in advance, and the acceleration can be adjusted automatically with any varying load. The acceleration is high at light load for fast startup and is low at heavy load. During speed up stage, the current amplitude  $I_m$  is large and constant for a large output torque. During constant speed stage, the current amplitude  $I_m$  is adjusted with load for high efficiency. After the algorithm runs for 0.1 s, the estimated angle error  $\theta_{\text{err}}^e$  is very close to its actual value  $\theta_{\text{err}}$ , which verifies the accuracy of angle error estimation (39). Throughout the whole process, the angle error  $\theta_{\text{err}}$  is near zero, which validates the effectiveness of this current angle controller. Accordingly, the  $q$ -axis current  $i_q$  is close to current amplitude  $I_m$  and the  $d$ -axis current  $i_d$  is relatively small especially in constant speed, which means  $i_d = 0$  control is achieved. In addition, the experimental waveforms in Fig. 20 are basically consistent with the simulation results in Fig. 16, which validates the precision of the large signal model.

### C. Transition From $I$ - $f$ Startup to FOC

In this experiment, the motor starts at rated load and then maintains a constant speed of 400 r/min. When the motor reaches its steady state operation, the control strategy switches from  $I$ - $f$  startup to FOC as illustrated in Fig. 21. At the switching moment 1.25 s, the  $d$ -axis current  $i_d$  decreases from zero to a negative value because the  $i_d = 0$  operation under  $I$ - $f$  startup switches to MTPA operation under sensorless FOC. However, the current amplitude  $I_m$  decreases slightly and the speed remains basically unchanged, which means the switching transition process from  $I$ - $f$  startup strategy to sensorless FOC is pretty smooth with zero angle error  $\theta_{\text{err}}$ .

In addition, this transition strategy is simple: First, set the initial reference torque of the speed loop output in the FOC strategy to the output torque of  $I$ - $f$  control. Second, adopt the  $dq$ -axis current references based on reference torque and MTPA strategy rather than  $I$ - $f$  control. Third, use the angle estimated by the sensorless method to replace the current vector angle given in the  $I$ - $f$  startup strategy.

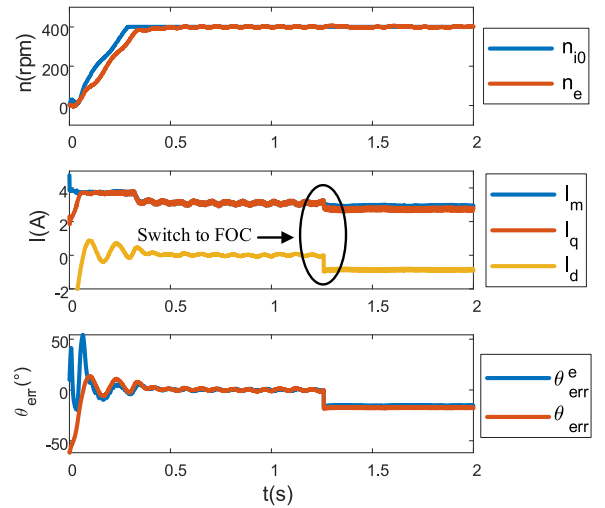


Fig. 21. Experimental results of transition from  $I$ - $f$  startup to FOC at rated load.

### D. Robustness Test

According to previous analysis, the electrical parameters used in the proposed  $I$ - $f$  startup strategy for estimating angle error only contain flux linkage  $\lambda_f$  and  $q$ -axis inductance  $L_q$ . As these parameters vary, the sensitivity tests are conducted to demonstrate the robustness of proposed strategy. When the motor starts at full load, the current amplitude is larger, the motor is easier to lose synchronization, i.e., the angle error range of successful startup becomes smaller according to Fig. 17. Therefore, the robustness tests are carried out at rated load.

The experimental waveforms at rated load under different estimated flux linkages are captured in Fig. 22. Regardless when  $\lambda_f^e = 50\%\lambda_f$  or  $\lambda_f^e = 150\%\lambda_f$ , the motor is able to start smoothly. In order to analyze the effect of the estimated flux linkage  $\lambda_f^e$  on estimated angle error  $\theta_{\text{err}}^e$ , only the time after the algorithm convergence, i.e., the period  $t > 0.1$  s is considered. When  $\lambda_f^e = 50\%\lambda_f$ , according to (44), the angle error  $\theta_{\text{err}} = 0.5\theta_{\text{err}}^e$ , so the absolute value of angle error  $|\theta_{\text{err}}|$  is less

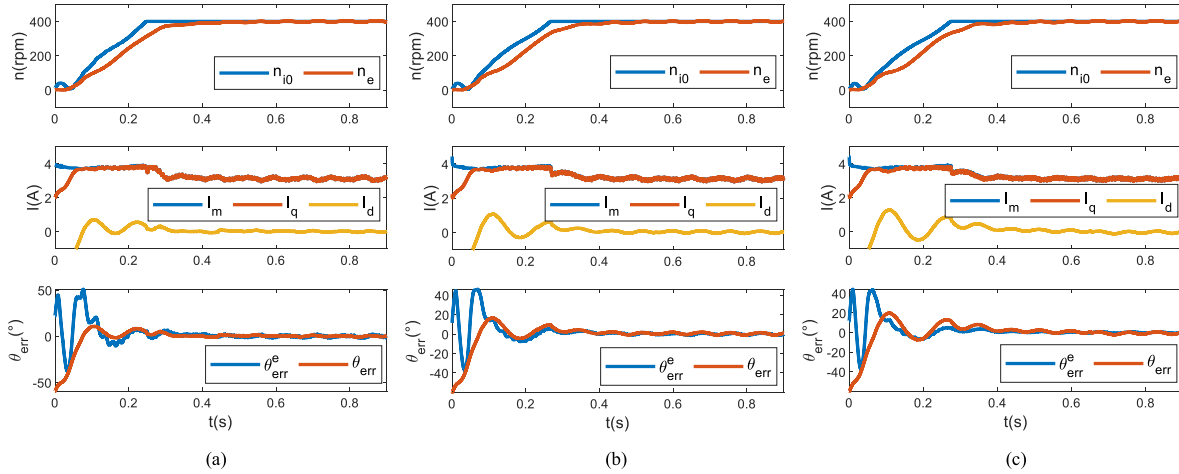


Fig. 22. Experimental results at rated load under different estimated flux linkages. From top to bottom: speed; current amplitude; and angle error. (a)  $\lambda_f^e = 50\%\lambda_f$ . (b)  $\lambda_f^e = 100\%\lambda_f$ . (c)  $\lambda_f^e = 150\%\lambda_f$

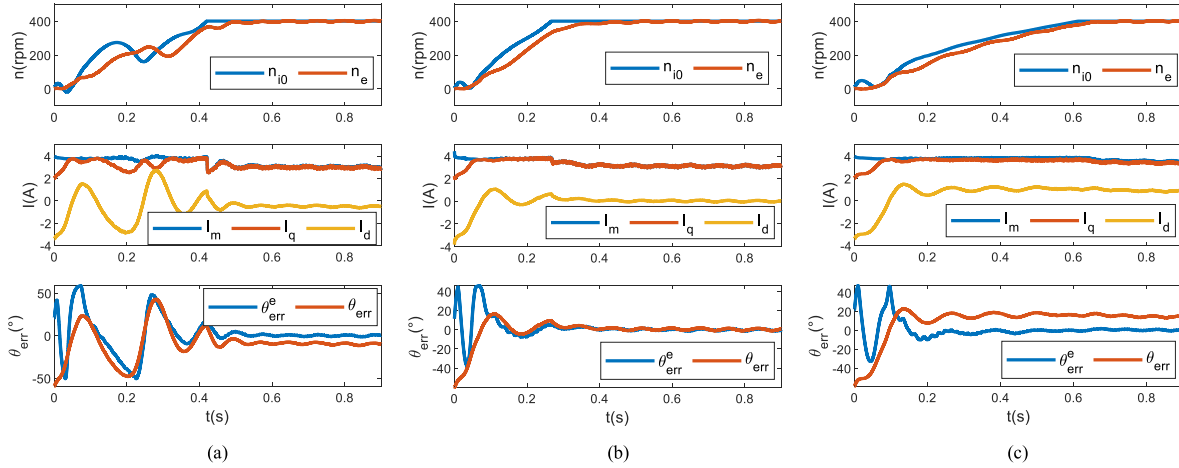


Fig. 23. Experimental results at rated load under different estimated  $q$ -axis inductances. From top to bottom: speed; current amplitude; and angle error. (a)  $L_q^e = 70\% L_q$ . (b)  $L_q^e = 100\% L_q$ . (c)  $L_q^e = 130\% L_q$

than the estimated absolute value  $|\theta_{\text{err}}^e|$  as shown in Fig. 22(a). Similarly, when  $\lambda_f^e = 150\%\lambda_f$ , the angle error  $\theta_{\text{err}} = 1.5\theta_{\text{err}}^e$  and in Fig. 22(c)  $|\theta_{\text{err}}|$  is greater than  $|\theta_{\text{err}}^e|$ . In other words, when the estimated flux linkage is not accurate, the actual angle error  $\theta_{\text{err}}$  is proportional to its estimated value. It is equivalent to the case where the gain of angle controller changes. As long as the current angle controller is reasonably designed and the system phase margin is relatively large, the system can work properly, even when the estimated flux linkage  $\lambda_f^e$  varies in the range of  $\pm 50\%$ .

For different estimated  $q$ -axis inductance  $L_q^e$  at rated load, the experimental results are presented in Fig. 23. Regardless of when  $L_q^e = 70\% L_q$  or  $L_q^e = 130\% L_q$ , the motor starts successfully. According to (45), the angle error  $\theta_{\text{err}} = \theta_{\text{err}}^e \mp \frac{0.3L_q I_\delta}{\lambda_f}$ , so there is a dc bias between the actual angle error  $\theta_{\text{err}}$  and its estimated value  $\theta_{\text{err}}^e$  in the experiments. The proposed current angle controller is able to make the estimated angle error  $\theta_{\text{err}}^e$  near zero, so the steady-state angle error is primarily  $\theta_{\text{err}} = \mp \frac{0.3L_q I_\delta}{\lambda_f}$ . According to above analysis (48), as long as the angle error  $\theta_{\text{err}}$

is greater than MTPA angle  $\gamma$  and smaller than load angle  $\theta_T$ , the motor is able to start smoothly. Besides, when  $L_q^e < L_q$ , the steady-state angle error  $\theta_{\text{err}}$  is less than zero and moves towards MTPA angle  $\gamma$ , reducing the angle margin and thus causing slight oscillation as shown in Fig. 23(a). When  $L_q^e > L_q$ , the angle error  $\theta_{\text{err}}$  is larger than zero and moves towards load angle  $\theta_T$ , reducing the output torque and thus increasing the startup time as shown in Fig. 23(c). The analysis is consistent with Fig. 17.

In summary, the proposed  $I$ - $f$  startup strategy has strong robustness in the presence of electrical parameters variation including flux linkage  $\lambda_f$  and  $q$ -axis inductance  $L_q$ .

## VII. CONCLUSION

In this article, an efficient  $I$ - $f$  control with large startup torque is introduced by autoadjusting the current vector acceleration first and then amplitude with varying load using the proposed current angle controller. The proposed angle controller input is the angle error between current vector and  $q$ -axis, which can be easily obtained through current oriented voltage equation.

Furthermore, the large-signal motion model is derived for evaluating  $I$ - $f$  startup performances, while the small-signal motion model is derived for designing controller parameters. The proposed current angle controller is validated through experimentation. The advantages are summarized as follows.

- 1) During speed up stage, the proposed angle controller adjusts acceleration with any load automatically to maintain zero angle error, which ensures full-load startup with as large startup torque as possible, while maintaining system stability.
- 2) During constant speed stage, the proposed current angle controller reduces the current amplitude until the angle error reaches zero, which improves efficiency significantly and ensures subsequent smooth transition from  $I$ - $f$  startup to sensorless FOC.
- 3) The proposed  $I$ - $f$  control has strong robustness against parameters variation, achieving full-load startup capability with  $\pm 50\% \lambda_f$  and  $\pm 30\% L_q$  parameters variation.

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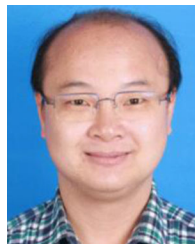


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